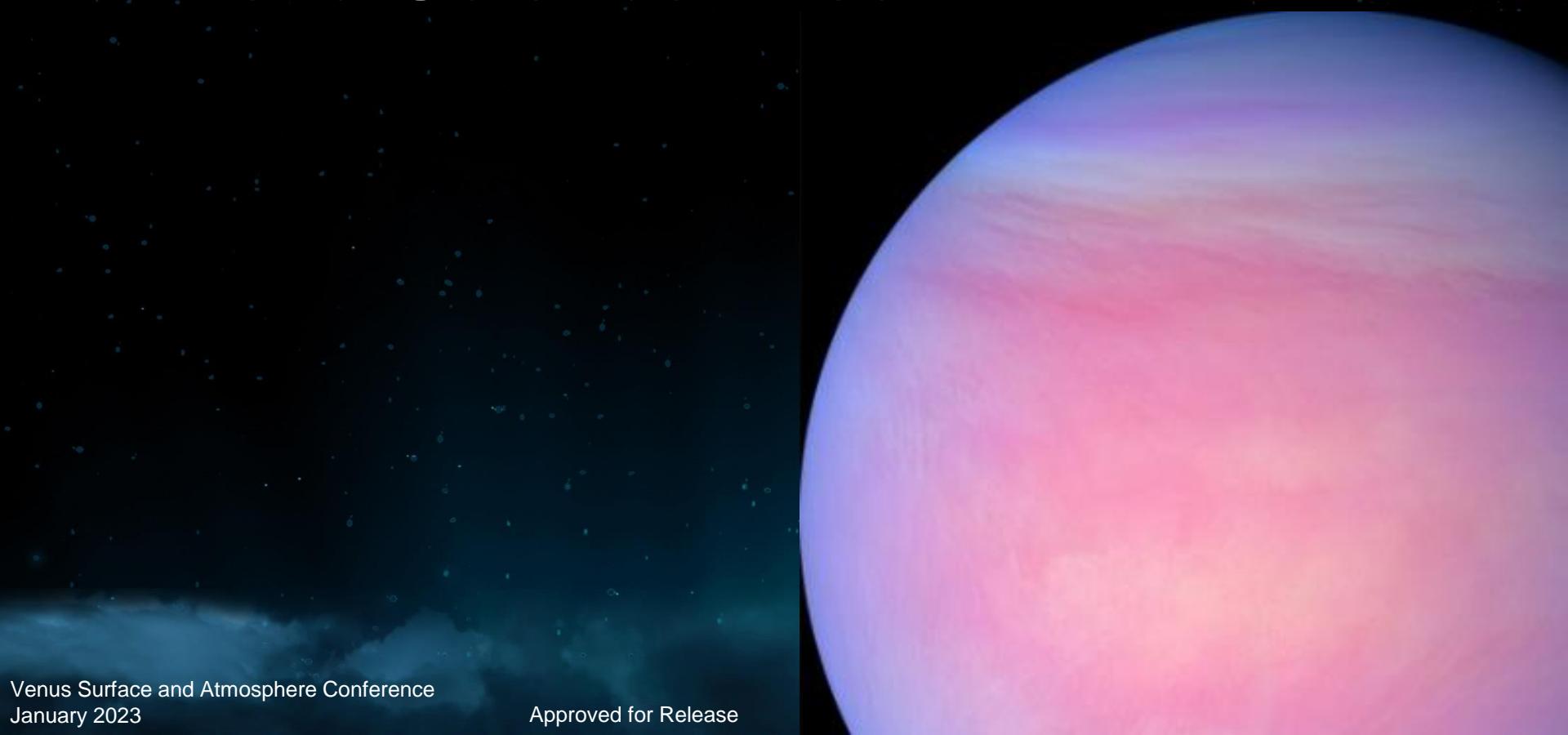


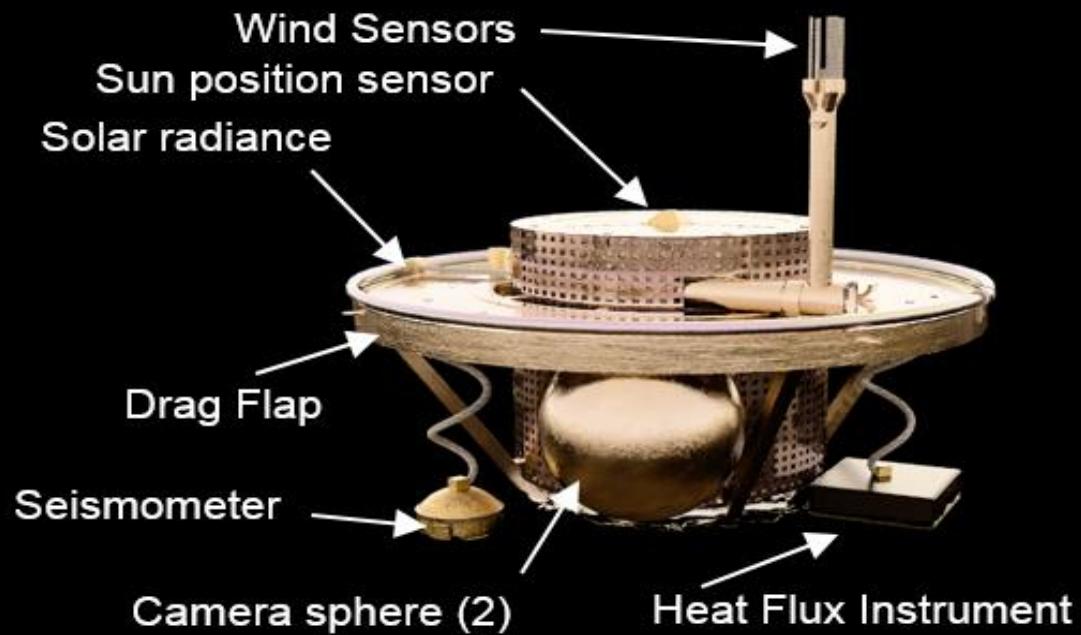
# Seismic and Atmospheric In situ Science Mission



# Outline

- Motivation and Challenges
- Long-Duration Lander Characteristics
- Status of System / Capability
- Next Steps

SAEVe study team: Richard Ghail, Martha Gilmore, Gary Hunter, Walter Kiefer, Sanjay Limaye, Michael Pauken, Colin Wilson, and Carol Tolbert



# Science Objectives

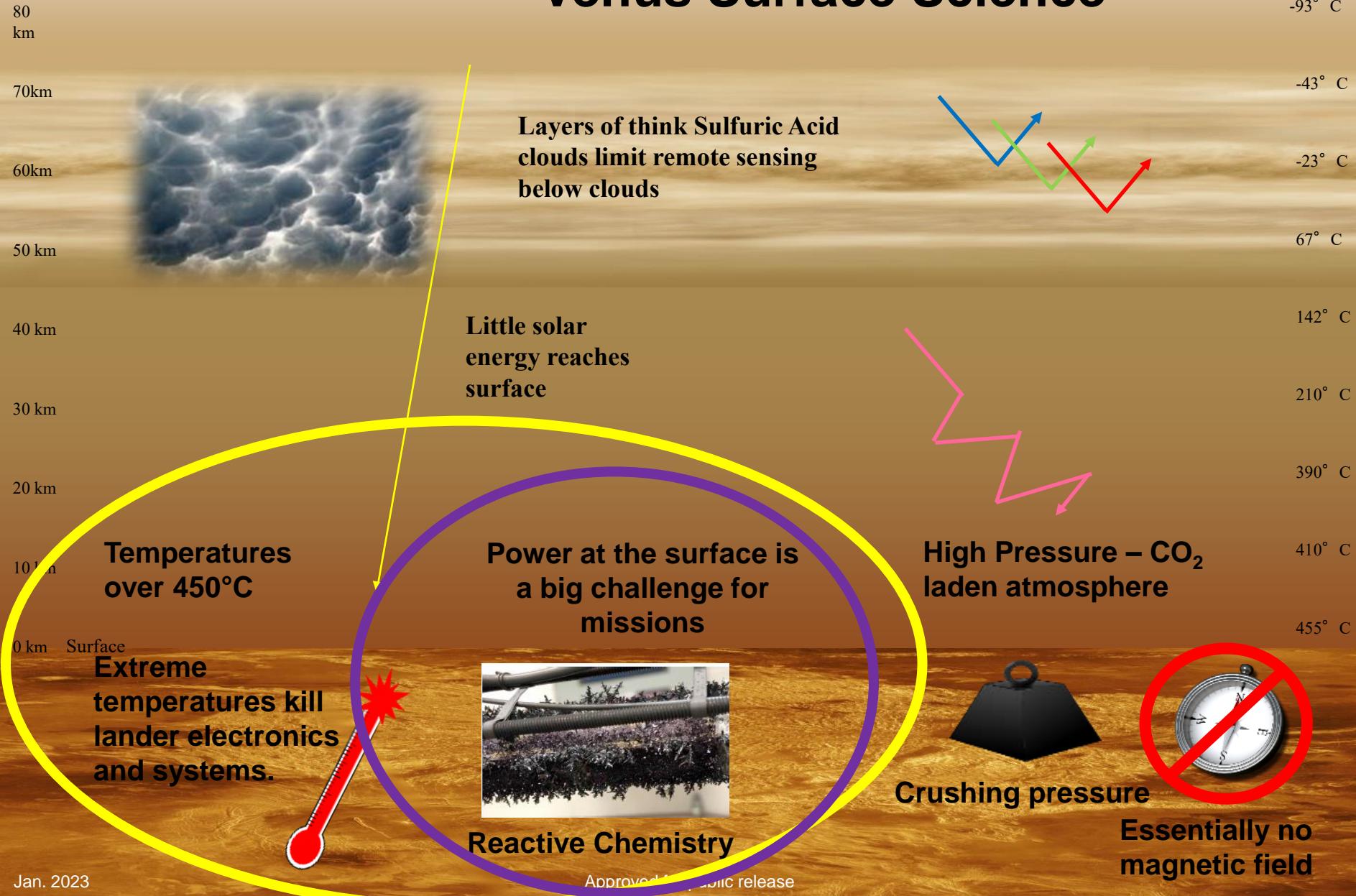
- We know why Venus is a compelling object of scientific interest
- SAEVe's unique strengths are acquiring ***In situ temporal data***
  - Venus seismicity, lithosphere structure and bulk composition
  - Meteorology, Chemistry variability
- SAEVe also contributes to rock and soil distribution and morphology, estimating momentum exchange, surface energy balance

2 Landers	<b>Science Objectives Tackled</b> <ol style="list-style-type: none"><li>1) Determine if Venus is seismically active and characterize the rate and style of activity,</li><li>2) Determine crust and lithosphere thickness and composition</li><li>3) Acquire temporal meteorological data to guide global circulation models</li><li>4) Estimate the momentum exchange between the planet and its atmosphere</li><li>5) Measure atmospheric chemistry variability</li><li>6) Determine current rate of heat loss from the Venus interior</li><li>7) Examine rock and soil distribution and morphology</li></ol>
One Lander	<b>Science Objectives Tackled</b> <ol style="list-style-type: none"><li>1) Determine if Venus is seismically active and characterize the rate and style of activity,</li><li>2) Acquire temporal meteorological data to guide global circulation models</li><li>3) Estimate the momentum exchange between the planet and its atmosphere</li><li>4) Measure atmospheric chemistry variability</li><li>5) Determine current rate of heat loss from the Venus interior</li><li>6) Examine rock and soil distribution and morphology</li></ol>
De-Scooped Lander	<b>Science Objectives Tackled</b> <ol style="list-style-type: none"><li>1) Determine if Venus is seismically active and characterize the rate and style of activity,</li><li>2) Acquire temporal meteorological data to guide global circulation models</li><li>3) Estimate the momentum exchange between the planet and its atmosphere</li><li>4) Measure atmospheric chemistry variability,</li></ol>



Entry challenges due to location  
and atmosphere of Venus

# The Technical Challenges of Venus Surface Science



# Lander Characteristics

- Technology advances are enabling operation of a small lander for 120 days or longer
  - SAEVe is based on capabilities developed by the LLISSE project
- Our approach is to design all lander components\* and subsystems to operate in the harsh Venus surface environment (T, P, Chemistry, Radiance, Seismicity).
- Lander is < 25 kg, operates for 120 days or more, and ~50cm at widest point

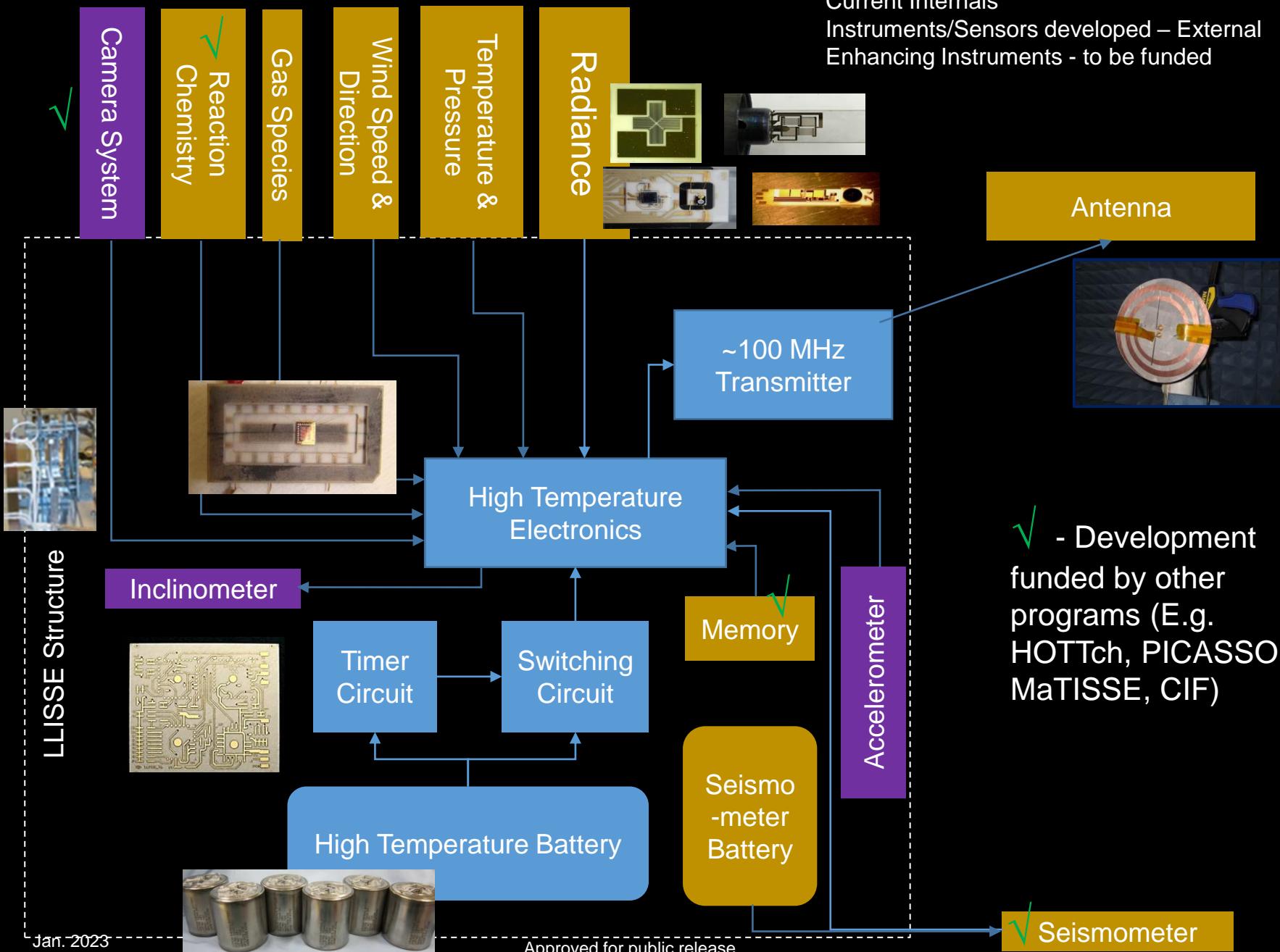
# SAEVe Block Diagram

## Legend

Current Internals

Instruments/Sensors developed – External

Enhancing Instruments - to be funded



# Top Level Status

- LLISSE has made tremendous progress on the capability to operate on the Venus surface
- SAEVe uses LLISSE subsystems and adds instruments and larger battery
- Status:
  - Robust high-temperature processing electronics, fabrication in process
  - High-temperature batteries
    - Full voltage battery tested at temp. and under simulated loads ~120 day life
    - Transitioned efforts toward packaging
  - Antenna material verification in process (just completing test in GEER)
  - Critical comm components in fabrication, will be assessed & fed back into design
  - High-temperature instruments & sensors
    - Varying readiness level – TRL 3 – 5+. Most at TRL4 or >
    - Seismometer in development under HOTTech-2
  - Operation of most sensors and core electronics in simulated Venus environment have been demonstrated several times
  - HEEET TPS assessed to deliver SAEVe to surface (HEEET at TRL-6 for Venus)

# Development Flow

Funded  
LLISSE  
Development

Communication

High-Temp Electronics T-5

Integrated sensors/electronics (T, P, Winds, Chem, Rad)

Power T-5

Structure T-5

LLISSE @ T-6

Demonstration test  
(life & performance)  
of full scale system in GEER

To Be Funded  
development  
related to  
Amaranth

Cameras

Prior Center funding. Next: PICASSO/ MatISSE, SAEVe direct or more Center funding

Seismometer

Current LLISSE and EPSCOR, Next: HOTTech-2, Direct Funding, PICASSO/MatISSE, or Center funding

Could be  
ready to start  
flight build  
~2028

Orbiter Related

Assessment not fully done (See slide for some possible development needs)

Legend:

DG-X  
T-X

Decision gate  
Expected TRL level

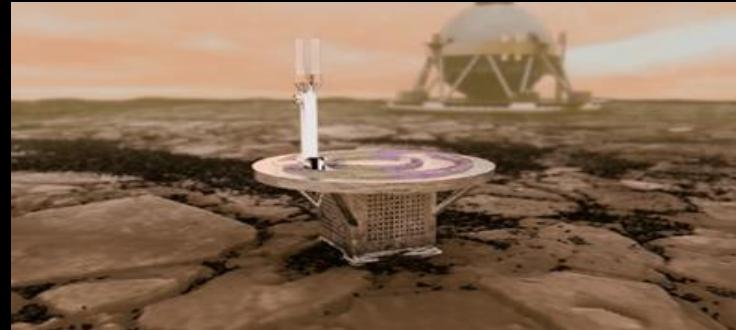
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GEER test (1 test can support multiple components)  
– Year TRL is estimated to be achieved

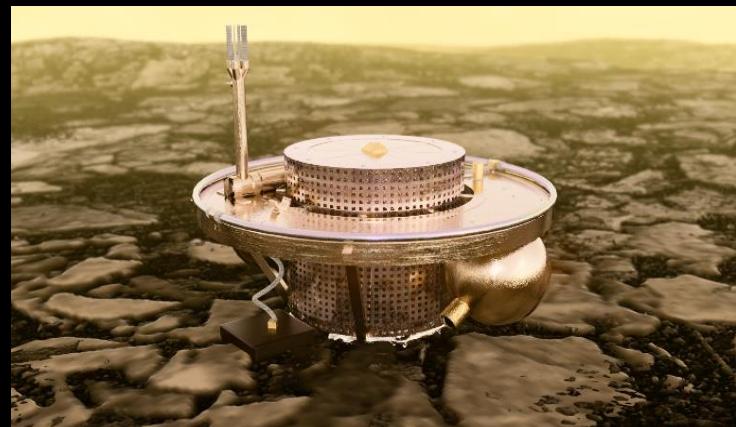


# Next-Steps / Take-aways

- Continue maturing technology and engage potential mission partners
  - Could support a mission late in 2020's, with resource availability
- Hardware exists for sub-systems and various levels of testing have been completed
- Major SAEVe instrument, the seismometer, also in development under HOTTech-2, low power memory as well



LLISSE Lander



SAEVe Concept

# Questions?

